Ballistic problems in the synthesis of orbital segment of satellite information system based on small and micro spacecraft

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The article presents an analytical review of the major ballistic problems in the synthesis of orbital segment of satellite information provision system based on small spacecraft. Much attention is paid to developing the concept of evaluating the possibility of universalization of small spacecraft satellite system for various purposes. On the basis of considering the problem statement of the synthesis of a ballistic satellite system structure as a Pareto-optimal system, it is shown that the small spacecraft system integration is possible primarily within one class of satellite systems or only "related" satellite system classes. Taking into consideration the trends in the development of classic heavy samples of space technology in civilian and dual-purpose systems, the requirements to technology precision of ballistic-navigation support for satellite system of spacecraft (including small spacecraft) for various purposes are formulated. The problems of using intersatellite navigation and ballistic aspects of the implementation of the inter-satellite measurement technologies are discussed. The transition from heavy spacecraft systems to small spacecraft satellite system requires solving totality of innovative problems associated with both the construction of the individual satellites, and with adaptive restructuring ballistic-navigation support and its systems.

Keywords: satellite system, information provision, small spacecraft, orbital segment synthesis, satellite system structure, system-ballistic characteristics, ballistic-navigation support, inter-satellite navigation measurements, local group (cluster), innovative technologies.

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